

IMPROVING COMPONENT LIFE AND PERFORMANCE

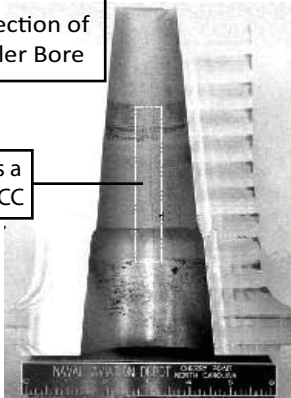
PROPELLER BLADES



LPB processing of an aluminum alloy propeller blade

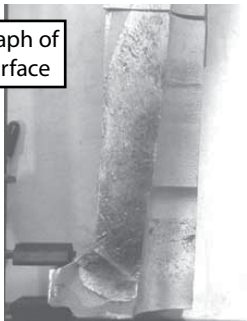
MITIGATING SCC IN PROPELLER BLADES

Cross Section of
Propeller Bore



Fracture as a
result of SCC

Macro fractograph of
the fracture surface

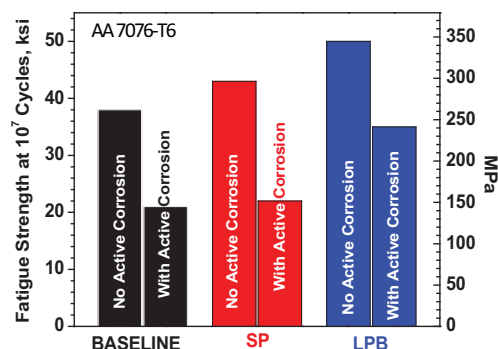


Low Plasticity Burnishing (LPB) mitigates stress corrosion cracking (SCC) and improves corrosion fatigue strength while increasing the service life and reducing the total maintenance costs of the aluminum alloy 7076-T6 propeller for the U.S. Navy's maritime patrol aircraft.

- Optimizes Maintenance Procedures
- Increases Time in Service
- Decreases Maintenance Costs
- Eliminates Scrapping

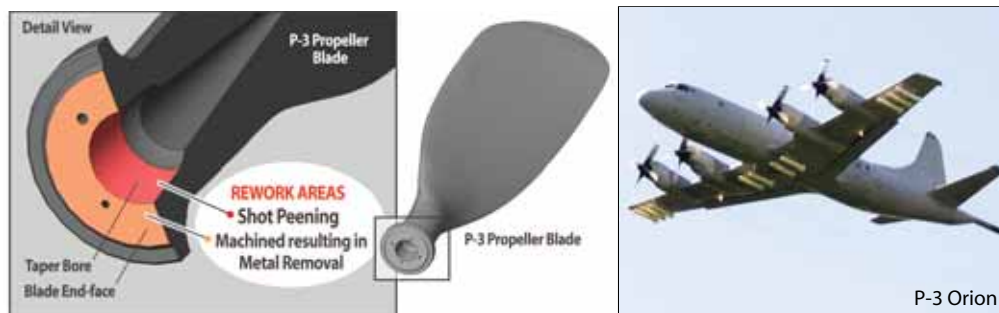
LPB eliminates premature replacement costs of the P-3 propeller taper bore by providing a deep layer of high magnitude compression of sufficient depth to mitigate SCC and improve corrosion fatigue.

Over time with environmental exposure, pitting corrosion occurs on the taper bore surface, creating the need to rework the bushing and taper bore area within the propeller blade. The pitting serves as a site for SCC and fatigue crack initiation. Shot peening is currently used to impart compressive residual stress and prolong the life of the taper bore. This leaves a dimpled surface which requires reaming of the bore to reduce the surface roughness and maximize the contact between the bronze bushing and the bore surface. Shot peening is also performed as a rework process at various stages, requiring further reaming of the taper bore. This shot peening and reaming process requires the blades to be scrapped every three cycles.



Corrosion fatigue, SCC and general corrosion tests conducted on aluminum alloy 7076-T6 demonstrated superior performance of the LPB process relative to the current shot peening and reaming process. Lambda Technologies developed the appropriate compressive residual stress distribution, performed corrosion and fatigue tests, designed and fabricated the tooling, LPB processed taper bores for component testing and prepared a production implementation plan.

LPB greatly reduces labor and material costs by eliminating the need for remachining and entirely removes the need to scrap blades. With LPB, turboprop engines can last years beyond their expected lifetimes, increasing time in service and decreasing the frequency of downtime due to maintenance, repair and inspection.



To learn how LPB can increase the life of your turboprop, please visit www.LambdaTechs.com or contact Kim Bellamy at (513) 561-0883.

References:

- D. Hornbach, G. Kauffman, T. Nguyen and P. Prevey, "Life Extension of the P-3 Aircraft by Robotic Applications of LPB." Aircraft, Airworthiness and Sustainment 2011. <http://www.meetingdata.utcd Dayton.com/agenda/airworthiness/2011/proceedings/presentations/P4619.pdf>. San Diego, CA, April 18-21, 2011.
- D. Hornbach, J. Scheel and P. Prevey, "Safe Life Conversion of Aircraft Aluminum Structures via Low Plasticity Burnishing for Mitigation of Corrosion Related Failures." DoD Corrosion 2009. Washington, D.C., August 10-14, 2009.

<http://www.lambdatechs.com/publications/all-technical-papers.html>

Accreditation:

- ISO/IEC 17025 Accredited Laboratory
- ISO 9001:2008 Certified
- FAA Accepted

For more information on Lambda, LPB® or to read complete papers, please visit www.LambdaTechs.com